

**PHY 711 Classical Mechanics and
Mathematical Methods
10-10:50 AM MWF Olin 103**

Plan for Lecture 37

- 1. Continued discussion of the effects of viscosity in fluid motion (Chap. 12)**
 - a. Comment on Stokes' viscosity**
 - b. Navier-Stokes equation**
- 2. Comments on Exam 2**

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23	Mon, 10/20/2014	Appendix A	Fourier transforms	#16
24	Wed, 10/22/2014	Chap. 5	Motion of Rigid Bodies	#17
25	Fri, 10/24/2014	Chap. 5	Motion of Rigid Bodies	#18
26	Mon, 10/27/2014	Chap. 5	Symmetric top in gravitational field	#18
27	Wed, 10/29/2014	Chap. 8	Vibrations of membranes	#19
28	Fri, 10/31/2014	Chap. 9	Physics of fluids	#20
29	Mon, 11/03/2014	Chap. 9	Physics of fluids	#21
30	Wed, 11/05/2014	Chap. 9	Sound waves	
31	Fri, 11/07/2014	Chap. 9	Sound waves	Begin Take-Home
32	Mon, 11/10/2014	Chap. 9	Non-linear effects	Continue Take-Home
33	Wed, 11/12/2014	Chap. 10	Surface waves in fluids	Continue Take-Home
34	Fri, 11/14/2014	Chap. 10	Surface waves in fluids	Continue Take-Home
35	Mon, 11/17/2014	Chap. 11	Heat Conduction	Take-Home due #22
36	Wed, 11/19/2014	Chap. 12	Viscosity	#23
37	Fri, 11/21/2014	Chap. 12	More viscosity	#24
38	Mon, 11/24/2014	Chap. 13	Elastic Continua	Prepare presentations
	Wed, 11/26/2014		Thanksgiving Holiday	
	Fri, 11/28/2014		Thanksgiving Holiday	
39	Mon, 12/01/2014	Chap. 13	Elastic Continua	Prepare presentations
	Wed, 12/03/2014		Student presentations I	
	Fri, 12/05/2014		Student presentations II	
	Mon, 12/08/2014		Begin Take-home final	Final grades due 12/17/2014

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Signup for PHY 711 Presentations

Wed. December 3, 2014

Time	Name	Title
10:00 AM	Lauren Nelson	
10:25 AM		

Fri. December 5, 2014

Time	Name	Title
10:00 AM		
10:25 AM		

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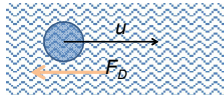
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Comments on Stokes' equation for viscous drag

Stokes' analysis of viscous drag on a sphere of radius R moving at speed u in medium with viscosity η :

$$F_D = -\eta(6\pi Ru)$$



Plan:

1. Consider the general effects of viscosity on fluid equations
2. Consider the solution to the linearized equations for the case of steady-state flow of a sphere of radius R
3. Infer the drag force needed to maintain the steady-state flow

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Newton - Euler equation for incompressible fluid, modified by viscous contribution (Navier - Stokes equation)

$$\frac{\partial \mathbf{v}}{\partial t} + (\mathbf{v} \cdot \nabla) \mathbf{v} = \mathbf{f}_{\text{applied}} - \frac{\nabla p}{\rho} + \frac{\eta}{\rho} \nabla^2 \mathbf{v}$$

Continuity equation : $\nabla \cdot \mathbf{v} = 0$ Irrotational flow : $\nabla \times \mathbf{v} = 0$

Assume steady state : $\Rightarrow \frac{\partial \mathbf{v}}{\partial t} = 0$

Assume non - linear effects small

Initially set $\mathbf{f}_{\text{applied}} = 0$; $\nabla p = \eta \nabla^2 \mathbf{v}$

Assume $\mathbf{v} = \nabla \times (\nabla \times f(r) \mathbf{u}) + \mathbf{u}$

where $f(r) \xrightarrow{r \rightarrow \infty} 0$

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$$\mathbf{v} = \nabla \times (\nabla \times f(r) \mathbf{u}) + \mathbf{u}$$

$$\mathbf{u} = u \hat{\mathbf{z}}$$

$$\nabla \times (\nabla \times f(r) \hat{\mathbf{z}}) = \nabla (\nabla \cdot f(r) \hat{\mathbf{z}}) - \nabla^2 f(r) \hat{\mathbf{z}}$$

$$\nabla \times \mathbf{v} = 0 \quad \Rightarrow \quad \nabla^2 (\nabla \times \mathbf{v}) = 0$$

$$\nabla^4 (\nabla \times f(r) \hat{\mathbf{z}}) = 0 \quad \Rightarrow \quad \nabla^4 (\nabla f(r) \times \hat{\mathbf{z}}) = 0 \quad \Rightarrow \quad \nabla^4 f(r) = 0$$

$$f(r) = C_1 r^2 + C_2 r + C_3 + \frac{C_4}{r}$$

$$v_r = u \cos \theta \left(1 - \frac{2}{r} \frac{df}{dr} \right) = u \cos \theta \left(1 - 4C_1 - \frac{2C_2}{r} - \frac{2C_4}{r^3} \right)$$

$$v_\theta = -u \sin \theta \left(1 - \frac{d^2 f}{dr^2} - \frac{1}{r} \frac{df}{dr} \right) = -u \sin \theta \left(1 - 4C_1 - \frac{C_2}{r} + \frac{C_4}{r^3} \right)$$

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$$v_r = u \cos \theta \left(1 - \frac{2}{r} \frac{df}{dr} \right) = u \cos \theta \left(1 - 4C_1 - \frac{2C_2}{r} - \frac{2C_4}{r^3} \right)$$

$$v_r = u \cos \theta \left(1 - \frac{2}{r} \frac{df}{dr} \right) = -u \sin \theta \left(1 - 4C_1 - \frac{C_2}{r} + \frac{C_4}{r^3} \right)$$

To satisfy $\mathbf{v}(r \rightarrow \infty) = \mathbf{u} : \Rightarrow C_1 = 0$

To satisfy $\mathbf{v}(R) = 0$ solve for C_2, C_4

$$v_r = u \cos \theta \left(1 - \frac{3R}{2r} + \frac{R^3}{2r^3} \right)$$

$$v_\theta = -u \sin \theta \left(1 - \frac{3R}{4r} - \frac{R^3}{4r^3} \right)$$

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$$v_r = u \cos \theta \left(1 - \frac{3R}{2r} + \frac{R^3}{2r^3} \right)$$

$$v_\theta = -u \sin \theta \left(1 - \frac{3R}{4r} - \frac{R^3}{4r^3} \right)$$

Determining pressure :

$$\nabla p = \eta \nabla^2 \mathbf{v} = -\eta \nabla \left(u \cos \theta \left(\frac{3R}{2r^2} \right) \right)$$

$$\Rightarrow p = p_0 - \eta u \cos \theta \left(\frac{3R}{2r^2} \right)$$

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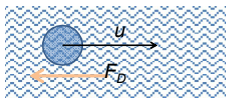
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$$p = p_0 - \eta u \cos \theta \left(\frac{3R}{2r^2} \right)$$

Corresponds to:

$$F_D = (p(R) - p_0) \frac{4\pi R^2}{\cos \theta}$$

$$= -\eta u (6\pi R)$$



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More general formulation of viscosity from Chapter 12 of Fetter and Walecka

For a non-viscous fluid, we can define a stress tensor:

$$T_{kl}^{ideal} = \rho v_k v_l + p \delta_{kl}$$

From Euler and Newton equations for fluid:

$$\int_V d^3r \frac{\partial(\rho v_k)}{\partial t} = - \sum_{l=1}^3 \int_A dA_l T_{kl} + \int_V d^3r \rho f_k$$

$$- \sum_{l=1}^3 \int_A dA_l T_{kl} = k\text{th component of force acting on surface } dA$$

For an ideal (non-viscous) fluid: $T_{kl} = T_{lk}$

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Effects of viscosity

Argue that viscosity is due to shear forces in a fluid of the form:

$$F_{drag} = \eta \frac{\partial v_x}{\partial y} dA$$

Formulate viscosity stress tensor with traceless and diagonal terms:

$$T_{kl}^v = -\eta \left(\frac{\partial v_k}{\partial x_l} + \frac{\partial v_l}{\partial x_k} - \frac{2}{3} \delta_{kl} (\nabla \cdot \mathbf{v}) \right) - \zeta \delta_{kl} (\nabla \cdot \mathbf{v})$$

↑ viscosity
 ↑ bulk viscosity

Total stress tensor: $T_{kl} = T_{kl}^{ideal} + T_{kl}^v$

$$T_{kl}^{ideal} = \rho v_k v_l + p \delta_{kl}$$

$$T_{kl}^v = -\eta \left(\frac{\partial v_k}{\partial x_l} + \frac{\partial v_l}{\partial x_k} - \frac{2}{3} \delta_{kl} (\nabla \cdot \mathbf{v}) \right) - \zeta \delta_{kl} (\nabla \cdot \mathbf{v})$$

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Effects of viscosity -- continued

Incorporating generalized stress tensor into Newton-Euler equations

$$\frac{\partial \rho v_i}{\partial t} + \sum_{j=1}^3 \frac{\partial \rho v_i v_j}{\partial x_j} = \rho f_i - \frac{\partial p}{\partial x_i} + \eta \sum_{j=1}^3 \frac{\partial^2 v_i}{\partial x_j^2} + \left(\zeta + \frac{1}{3} \eta \right) \sum_{j=1}^3 \frac{\partial^2 v_j}{\partial x_i \partial x_j}$$

Continuity condition

$$\frac{\partial \rho}{\partial t} + \sum_{j=1}^3 \frac{\partial \rho v_j}{\partial x_j} = 0$$

Vector form (Navier-Stokes equation)

$$\frac{\partial \mathbf{v}}{\partial t} + (\mathbf{v} \cdot \nabla) \mathbf{v} = \mathbf{f} - \frac{1}{\rho} \nabla p + \frac{\eta}{\rho} \nabla^2 \mathbf{v} + \frac{1}{\rho} \left(\zeta + \frac{1}{3} \eta \right) \nabla (\nabla \cdot \mathbf{v})$$

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Comments on Exam 2

1. Consider a displacement function $u(x, t)$ representing a one-dimensional traveling wave (either transverse or longitudinal) which is a solution of the one-dimensional wave equation with wave speed c :

$$\frac{\partial^2 u}{\partial x^2} - \frac{1}{c^2} \frac{\partial^2 u}{\partial t^2} = 0.$$

If the initial conditions for the wave displacement $u(x, t)$ are given by

$$u(x, 0) = U_0 e^{-(x-x_0)^2/\sigma^2},$$

and

$$\frac{\partial u}{\partial t}(x, 0) = V_0 \left(\frac{x}{\mu}\right)^3 e^{-(x/\mu)^4},$$

find the form of $u(x, t)$ for $t > 0$. Express your result in terms of the constants U_0 , V_0 , σ , μ , x_0 , and c .

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Solution for $u(x, 0) \equiv \varphi(x)$ and $\frac{\partial u}{\partial t}(x, 0) \equiv \psi(x)$:

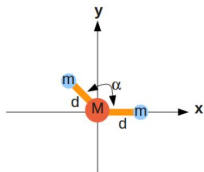
$$\Rightarrow u(x, t) = \frac{1}{2}(\varphi(x - ct) + \varphi(x + ct)) + \frac{1}{2c} \int_{x-ct}^{x+ct} \psi(x') dx'$$

$$u(x, t) = \frac{U_0}{2} \left(e^{-(x-x_0+ct)/\sigma^2} + e^{-(x-x_0-ct)/\sigma^2} \right) + \frac{V_0 \mu}{8c} \left(e^{-((x+ct)/\mu)^4} - e^{-((x-ct)/\mu)^4} \right)$$

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2.

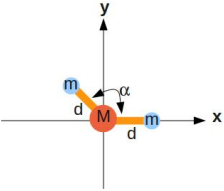
The figure above shows a triatomic molecule aligned as shown with $M = 16m$ at the origin and the other two masses m in the $x - y$ plane at the locations $(d\hat{x})$ and $(d\cos(\alpha)\hat{x} + d\sin(\alpha)\hat{y})$.

- Find the moment of inertial tensor for the molecule in the given coordinate system, expressing your answer in terms of the parameters m , d , and α .
- Find the principal moments of inertia and the corresponding principal axes for this coordinate system.
- Find the center of mass of this system.
- Find the principal moments of inertia and corresponding principal axes for rotating this system about its center of mass.

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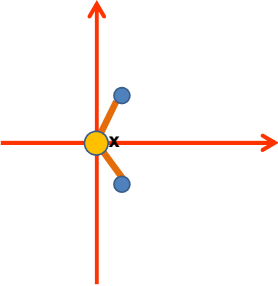
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$$I = md^2 \begin{pmatrix} \sin^2 \alpha & -\sin \alpha \cos \alpha & 0 \\ -\sin \alpha \cos \alpha & 1 + \cos^2 \alpha & 0 \\ 0 & 0 & 2 \end{pmatrix}$$

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Principal moments of inertia



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The 3 Euler angles, α , β , and γ were defined as a series of rotations to take a vector $\mathbf{v}$ in one coordinate frame to another frame in which is written as $\mathbf{v}'$. Alternatively, these angles can represent a general rotation of a vector quantity within a given coordinate frame.

$$\mathbf{v}' = \mathbf{R}\mathbf{v} \quad \text{where} \quad \mathbf{R} = \mathbf{R}_\alpha \mathbf{R}_\beta \mathbf{R}_\gamma$$

It can be shown that the 3×3 matrix components of the rotation matrix are given by:

$$\begin{aligned} R_{xx} &= \cos \alpha \cos \beta \cos \gamma - \sin \alpha \sin \gamma \\ R_{xy} &= \sin \alpha \cos \beta \cos \gamma + \cos \alpha \sin \gamma \\ R_{xz} &= -\sin \beta \cos \gamma \\ R_{yx} &= -\cos \alpha \cos \beta \sin \gamma - \sin \alpha \cos \gamma \\ R_{yy} &= -\sin \alpha \cos \beta \sin \gamma + \cos \alpha \cos \gamma \\ R_{yz} &= \sin \beta \sin \gamma \\ R_{zx} &= \cos \alpha \sin \beta \\ R_{zy} &= \sin \alpha \sin \beta \\ R_{zz} &= \cos \beta \end{aligned}$$

As an example, consider the vector

$$\mathbf{v} = \begin{pmatrix} \sqrt{1/3} \\ \sqrt{1/3} \\ \sqrt{1/3} \end{pmatrix}$$

(a) Evaluate the matrix $\mathbf{R}$ and find the vector $\mathbf{v}' = \mathbf{R}\mathbf{v}$ when $\alpha = \pi/4$, $\cos(\beta) = \sqrt{1/3}$, and $\gamma = -\pi/3$.

(b) Evaluate the matrix $\mathbf{R}$ and find the vector $\mathbf{v}' = \mathbf{R}\mathbf{v}$ when $\alpha = \pi/4$, $\cos(\beta) = \sqrt{1/3}$, and $\gamma = \pi/3$.

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$$\text{a) } \mathcal{R} = \begin{pmatrix} \sqrt{\frac{2}{3}} & -\sqrt{\frac{1}{6}} & -\sqrt{\frac{1}{6}} \\ 0 & \sqrt{\frac{1}{2}} & -\sqrt{\frac{1}{2}} \\ \sqrt{\frac{1}{3}} & \sqrt{\frac{1}{3}} & \sqrt{\frac{1}{3}} \end{pmatrix} \quad \mathbf{v}' = \mathcal{R}\mathbf{v} = \begin{pmatrix} 0 \\ 0 \\ 1 \end{pmatrix}$$

$$\text{b) } \mathcal{R} = \begin{pmatrix} -\sqrt{\frac{1}{6}} & \sqrt{\frac{2}{3}} & -\sqrt{\frac{1}{6}} \\ -\sqrt{\frac{1}{2}} & 0 & \sqrt{\frac{1}{2}} \\ \sqrt{\frac{1}{3}} & \sqrt{\frac{1}{3}} & \sqrt{\frac{1}{3}} \end{pmatrix} \quad \mathbf{v}' = \mathcal{R}\mathbf{v} = \begin{pmatrix} 0 \\ 0 \\ 1 \end{pmatrix}$$

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Consider linear sound waves in a gas confined between two rigid concentric spheres of radii a and b . The velocity potential $\Phi(r, t)$ for the sound wave satisfies the equation

$$\frac{\partial^2 \Phi}{\partial t^2} - c^2 \nabla^2 \Phi = 0,$$

where c denotes the speed of sound (assumed to be constant). Because of the spherical geometry, it is convenient to factor the velocity potential into radial, angular, and time dependent terms:

$$\Phi(r, \theta, \phi, t) = f(r)Y_{lm}(\theta, \phi)e^{i\omega t},$$

where ω denotes a harmonic frequency and $Y_{lm}(\theta, \phi)$ denotes the angular function which satisfies the equation

$$\frac{1}{r^2} \left(\frac{\partial}{\partial \theta} \sin \theta \frac{\partial}{\partial \theta} + \frac{1}{\sin^2 \theta} \frac{\partial^2}{\partial \phi^2} \right) Y_{lm}(\theta, \phi) = -l(l+1)Y_{lm}(\theta, \phi).$$

(a) Find the equation that the radial functions $f(r)$ must satisfy to represent the radial part of the normal modes of the sound waves.

(b) Show that the boundary conditions for the normal mode functions are

$$\frac{df(r)}{dr} = 0 \quad \text{and} \quad \frac{df(r)}{dr} = 0.$$

(c) Numerically determine the lowest frequency spherically symmetric ($l=0$) normal mode. Find frequency ω as a multiple of the speed of sound c and the corresponding radial function.

(d) Numerically determine the lowest frequency $l=1$ normal mode. Find frequency ω as a multiple of the speed of sound c and the corresponding radial function.

Note - the following functional forms for the spherical Bessel functions may prove useful:

$$j_0(x) = \frac{\sin(x)}{x} \quad y_0(x) = -\frac{\cos(x)}{x}$$

$$j_1(x) = \frac{\sin(x)}{x^2} - \frac{\cos(x)}{x} \quad y_1(x) = -\frac{\cos(x)}{x^2} - \frac{\sin(x)}{x}$$

$$j_2(x) = \left(\frac{3}{x^3} - \frac{1}{x} \right) \sin(x) - \frac{3 \cos(x)}{x^2} \quad y_2(x) = \left(\frac{3}{x^3} + \frac{1}{x} \right) \cos(x) - \frac{3 \sin(x)}{x^2}$$

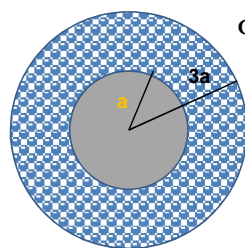
These spherical Bessel and Neumann functions ($j_l(x)$ for $x = j$ or $x = y$) satisfy the differential equation:

$$\left(\frac{d^2}{dx^2} + \frac{2}{x} \frac{d}{dx} - \frac{l(l+1)}{x^2} + 1 \right) \psi(x) = 0.$$

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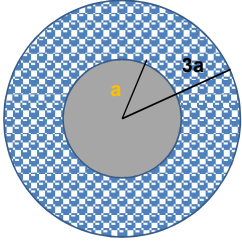
$$\Phi(\mathbf{r}, t) = \phi_l(r) Y_{lm}(\hat{\mathbf{r}}) e^{-i\omega t}$$

$$\phi_l(r) = C_l (j_l(kr) + x_l y_l(kr))$$

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$\varphi_l(r) = C_l (j_l(kr) + x_l y_l(kr))$
 Boundary values:
 $\frac{d\varphi_l(a)}{dr} = 0 = \frac{d\varphi_l(3a)}{dr}$

$j'_l(ka) + x_l y'_l(ka) = 0 = j'_l(3ka) + x_l y'_l(3ka)$
 $x_l = -\frac{j'_l(ka)}{y'_l(ka)} = -\frac{j'_l(3ka)}{y'_l(3ka)}$

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