

PHY 711 Classical Mechanics and Mathematical Methods 9-9:50 AM MWF Olin 107

Plan for Lecture 31: Chap. 9 of F&W

Wave equation for sound in the linear approximation

1. Sound scattering
2. Non-linear effects in sound waves

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22	Mon, 10/23/2017	Chap. 7	Solutions of Sturm-Liouville equations		
23	Wed, 10/25/2017	Chap. 7	Solutions of Sturm-Liouville equations		
24	Fri, 10/27/2017	App. A	Laplace transforms and contour integrals	#13	11/01/2017
25	Mon, 10/30/2017	App. A	Contour integrals		
26	Wed, 11/01/2017	Chap. 8	Mechanics of Elastic Membranes	#14	11/06/2017
27	Fri, 11/03/2017	Chap. 9	Introduction to hydrodynamics		
28	Mon, 11/06/2017	Chap. 9	Introduction to hydrodynamics	#15	11/10/2017
29	Wed, 11/08/2017	Chap. 9	Sound waves		
30	Fri, 11/10/2017	Chap. 9	Sound waves	#16	11/17/2017
	Mon, 11/13/2017		Class cancelled		
31	Wed, 11/15/2017	Chap. 9	Sound waves -- including non-linearities		
32	Fri, 11/17/2017	Chap. 10	Surface waves in fluids		
33	Mon, 11/20/2017	Chap. 10	Surface waves in fluids		
	Wed, 11/22/2017		Thanksgiving Holiday -- No class		
	Fri, 11/24/2017		Thanksgiving Holiday -- No class		
34	Mon, 11/27/2017	Chap. 11	Heat conductivity		
35	Wed, 11/29/2017	Chap. 12	Viscous fluids		
36	Fri, 12/01/2017	Chap. 12	Viscous fluids		
	Mon, 12/04/2017		Presentations I		
	Wed, 12/06/2017		Presentations II		
	Fri, 12/08/2017		Presentations III		

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Events

Colloquium: Nov. 15, 2017 at 4 PM
 WPU Physics Colloquium TITLE: "What is fossilization? Casting new light (and electrons) on an old question" SPEAKER: Professor Elizabeth Braden, Department of Engineering, Wake Forest University, Winston-Salem, NC TIME: Wed, ...

Colloquium: Nov. 29, 2017 at 4 PM
 WPU Physics Colloquium TITLE: "Cosmic Luminosity of Quasar Outflows and its Implications to Galaxy Formation" SPEAKER: Professor Nathan Arav, Department of Physics Virginia Tech, Blacksburg, VA TIME: Wed, Nov. 29, 2017.

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Solutions to wave equation:

$$\nabla^2 \Phi - \frac{1}{c^2} \frac{\partial^2 \Phi}{\partial t^2} = 0$$

Plane wave solution :

$$\Phi(\mathbf{r}, t) = A e^{i\mathbf{k} \cdot \mathbf{r} - i\omega t} \quad \text{where} \quad k^2 = \left(\frac{\omega}{c}\right)^2$$

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Scattering of sound waves –
for example, from a rigid cylinder

Figure 51.8 Scattering from a rigid cylinder.
Figure from Fetter and Walecka pg. 337

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Scattering of sound waves –
for example, from a rigid cylinder

Velocity potential --

$$\Phi(\mathbf{r}) = \Phi_{inc}(\mathbf{r}) + \Phi_{sc}(\mathbf{r}) \quad \Phi_{inc}(\mathbf{r}) = e^{i\mathbf{k} \cdot \mathbf{r}}$$

Helmholz equation in cylindrical coordinates:

$$(\nabla^2 + k^2)\Phi(\mathbf{r}) = 0 = \left(\frac{1}{r} \frac{\partial}{\partial r} r \frac{\partial}{\partial r} + \frac{1}{r^2} \frac{\partial^2}{\partial \phi^2} + \frac{\partial^2}{\partial z^2} + k^2 \right) \Phi(\mathbf{r})$$

Assume: $\Phi(\mathbf{r}) = \sum_{m=-\infty}^{\infty} e^{im\phi} R_m(r)$

where $\left(\frac{d^2}{dr^2} + \frac{1}{r} \frac{d}{dr} - \frac{m^2}{r^2} + k^2 \right) R_m(r) = 0$

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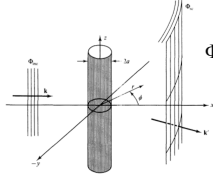


Figure 51.8 Scattering from a rigid cylinder.

$$\Phi_{inc}(\mathbf{r}) = e^{i\mathbf{k}\cdot\mathbf{r}} = e^{ikr \cos \phi} = \sum_{m=-\infty}^{\infty} i^m e^{im\phi} J_m(kr)$$

Note that for integer m , $J_m(z) = (-1)^m J_{-m}(z)$ so that this expression can be simplified.

$$\Phi_{sc}(\mathbf{r}) = \sum_{m=-\infty}^{\infty} C_m e^{im\phi} H_m(kr) \quad \text{where Hankel function}$$

represents an outgoing wave: $H_m(kr) = J_m(kr) + iN_m(kr)$

Boundary condition at $r = a$: $\left. \frac{\partial \Phi}{\partial r} \right|_{r=a} = 0$

$$\Rightarrow i^m J'_m(ka) + C_m H'_m(ka) = 0 \quad C_m = -i^m \frac{J'_m(ka)}{H'_m(ka)}$$

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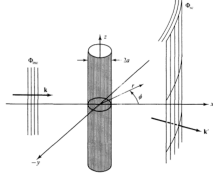


Figure 51.8 Scattering from a rigid cylinder.

$$\Phi_{sc}(\mathbf{r}) = - \sum_{m=-\infty}^{\infty} i^m \frac{J'_m(ka)}{H'_m(ka)} e^{im\phi} H_m(kr)$$

Asymptotic form:

$$i^m H_m(kr) \approx \sqrt{\frac{2}{\pi kr}} e^{i(kr - \pi/4)}$$

$$\Phi_{sc}(\mathbf{r}) \approx f(\phi) \sqrt{\frac{1}{r}} e^{ikr} = - \sum_{m=-\infty}^{\infty} \frac{J'_m(ka)}{H'_m(ka)} e^{im\phi} \sqrt{\frac{2}{\pi kr}} e^{i(kr - \pi/4)}$$

$$\Rightarrow f(\phi) = - \sqrt{\frac{2}{\pi k}} \sum_{m=-\infty}^{\infty} \frac{J'_m(ka)}{H'_m(ka)} e^{i(m\phi - \pi/4)}$$

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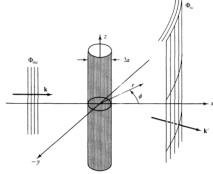
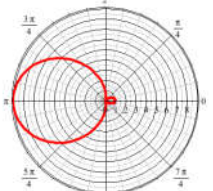


Figure 51.8 Scattering from a rigid cylinder.

$$\frac{d\sigma}{d\phi} = |f(\phi)|^2$$

$$f(\phi) = - \sqrt{\frac{2}{\pi k}} \sum_{m=-\infty}^{\infty} \frac{J'_m(ka)}{H'_m(ka)} e^{i(m\phi - \pi/4)}$$

For $ka \ll 1$

$$\frac{d\sigma}{d\phi} = |f(\phi)|^2 \approx \frac{1}{8} \pi k^3 a^4 (1 - 2 \cos \phi)^2$$


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Effects of nonlinearities in fluid equations
 -- one dimensional case
 Newton - Euler equation of motion :

$$\frac{\partial \mathbf{v}}{\partial t} + (\mathbf{v} \cdot \nabla) \mathbf{v} = \mathbf{f}_{\text{applied}} - \frac{\nabla p}{\rho}$$

Continuity equation : $\frac{\partial \rho}{\partial t} + \nabla \cdot (\rho \mathbf{v}) = 0$

Assume spatial variation confined to x direction ;
 assume that $\mathbf{v} = v \hat{\mathbf{x}}$ and $\mathbf{f}_{\text{applied}} = 0$.

$$\frac{\partial v}{\partial t} + v \frac{\partial v}{\partial x} = -\frac{1}{\rho} \frac{\partial p}{\partial x}$$

$$\frac{\partial \rho}{\partial t} + v \frac{\partial \rho}{\partial x} + \rho \frac{\partial v}{\partial x} = 0$$

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$$\frac{\partial v}{\partial t} + v \frac{\partial v}{\partial x} + \frac{1}{\rho} \frac{\partial p}{\partial x} = 0$$

$$\frac{\partial \rho}{\partial t} + v \frac{\partial \rho}{\partial x} + \rho \frac{\partial v}{\partial x} = 0$$

Expressing p in terms of ρ : $p = p(\rho)$

$$\frac{\partial p}{\partial x} = \frac{\partial p}{\partial \rho} \frac{\partial \rho}{\partial x} \equiv c^2(\rho) \frac{\partial \rho}{\partial x} \quad \text{where} \quad \frac{\partial p}{\partial \rho} \equiv c^2(\rho)$$

For adiabatic ideal gas: $\frac{dp}{p} = \gamma \frac{d\rho}{\rho} \quad p = p_0 \left(\frac{\rho}{\rho_0} \right)^\gamma$

$$c^2(\rho) = \frac{\gamma p}{\rho} = c_0^2 \left(\frac{\rho}{\rho_0} \right)^{\gamma-1} \quad \text{where} \quad c_0^2 \equiv \frac{\gamma p_0}{\rho_0}$$

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$$\frac{\partial v}{\partial t} + v \frac{\partial v}{\partial x} + \frac{c^2(\rho)}{\rho} \frac{\partial \rho}{\partial x} = 0$$

$$\frac{\partial \rho}{\partial t} + v \frac{\partial \rho}{\partial x} + \rho \frac{\partial v}{\partial x} = 0$$

Expressing variation of v in terms of $v(\rho)$:

$$\frac{\partial v}{\partial \rho} \frac{\partial \rho}{\partial t} + v \frac{\partial v}{\partial \rho} \frac{\partial \rho}{\partial x} + \frac{c^2(\rho)}{\rho} \frac{\partial \rho}{\partial x} = 0$$

$$\frac{\partial \rho}{\partial t} + v \frac{\partial \rho}{\partial x} + \rho \frac{\partial v}{\partial \rho} \frac{\partial \rho}{\partial x} = 0$$

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Some more algebra :

$$\text{From Euler equation : } \frac{\partial v}{\partial \rho} \left(\frac{\partial \rho}{\partial t} + v \frac{\partial \rho}{\partial x} \right) + \frac{c^2(\rho)}{\rho} \frac{\partial \rho}{\partial x} = 0$$

$$\text{From continuity equation : } \frac{\partial \rho}{\partial t} + v \frac{\partial \rho}{\partial x} = -\rho \frac{\partial v}{\partial \rho} \frac{\partial \rho}{\partial x}$$

$$\text{Combined equation : } \frac{\partial v}{\partial \rho} \left(-\rho \frac{\partial v}{\partial \rho} \frac{\partial \rho}{\partial x} \right) + \frac{c^2(\rho)}{\rho} \frac{\partial \rho}{\partial x} = 0$$

$$\Rightarrow \left(\frac{\partial v}{\partial \rho} \right)^2 = \frac{c^2(\rho)}{\rho^2} \quad \frac{\partial v}{\partial \rho} = \pm \frac{c}{\rho}$$

$$\Rightarrow \frac{\partial \rho}{\partial t} + (v \pm c) \frac{\partial \rho}{\partial x} = 0$$

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$$\text{Assuming adiabatic process : } c^2 = c_0^2 \left(\frac{\rho}{\rho_0} \right)^{\gamma-1} \quad c_0^2 = \frac{\gamma p_0}{\rho_0}$$

$$\frac{\partial v}{\partial \rho} = \frac{dv}{d\rho} = \pm \frac{c}{\rho} \quad \Rightarrow v = \pm c_0 \int_{\rho_0}^{\rho} \left(\frac{\rho'}{\rho_0} \right)^{(\gamma-1)/2} \frac{d\rho'}{\rho'}$$

$$\Rightarrow v = \pm \frac{2c_0}{\gamma-1} \left(\left(\frac{\rho}{\rho_0} \right)^{(\gamma-1)/2} - 1 \right)$$

$$\Rightarrow c = c_0 \left(\frac{\rho}{\rho_0} \right)^{(\gamma-1)/2}$$

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Summary :

$$\frac{dv}{d\rho} = \pm \frac{c}{\rho}$$

$$\frac{\partial \rho}{\partial t} + (v \pm c) \frac{\partial \rho}{\partial x} = 0$$

$$\text{Assuming adiabatic process : } c^2 = c_0^2 \left(\frac{\rho}{\rho_0} \right)^{\gamma-1} \quad c_0^2 = \frac{\gamma p_0}{\rho_0}$$

$$c = c_0 \left(\frac{\rho}{\rho_0} \right)^{(\gamma-1)/2} \quad v = \pm \frac{2c_0}{\gamma-1} \left(\left(\frac{\rho}{\rho_0} \right)^{(\gamma-1)/2} - 1 \right)$$

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Traveling wave solution:

Assume: $\rho = \rho_0 + f(x - u(\rho)t)$

Need to find self-consistent equations for propagation velocity $u(\rho)$ using equations

From previous derivations: $\frac{\partial \rho}{\partial t} + (v \pm c) \frac{\partial \rho}{\partial x} = 0$

Apparently: $u(\rho) \Leftrightarrow v \pm c$

For adiabatic ideal gas and + signs:

$$u = v + c = c_0 \left(\frac{\gamma + 1}{\gamma - 1} \left(\frac{\rho}{\rho_0} \right)^{(\gamma - 1)/2} - \frac{2}{\gamma - 1} \right)$$

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Traveling wave solution -- continued:

$$\frac{\partial \rho}{\partial t} + (v \pm c) \frac{\partial \rho}{\partial x} = 0$$

Assume: $\rho = \rho_0 + f(x - u(\rho)t) = \rho_0 + f(x - (v \pm c)t)$

For adiabatic ideal gas and + signs:

$$u = v + c = c_0 \left(\frac{\gamma + 1}{\gamma - 1} \left(\frac{\rho}{\rho_0} \right)^{(\gamma - 1)/2} - \frac{2}{\gamma - 1} \right)$$

Solution in linear approximation:

$$u = v + c \approx v_0 + c_0 = c_0 \left(\frac{\gamma + 1}{\gamma - 1} - \frac{2}{\gamma - 1} \right) = c_0$$

$$\Rightarrow \rho = \rho_0 + f(x - c_0 t)$$

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Traveling wave solution -- full non-linear case:

Visualization for particular waveform: $\rho = \rho_0 + f(x - \underbrace{u(\rho)t}_w)$

Assume: $f(w) \equiv \rho_0 s(w)$

$$\frac{\rho}{\rho_0} = 1 + s(x - ut)$$

For adiabatic ideal gas:

$$u = c_0 \left(\frac{\gamma + 1}{\gamma - 1} \left(\frac{\rho}{\rho_0} \right)^{(\gamma - 1)/2} - \frac{2}{\gamma - 1} \right)$$

$$u = c_0 \left(\frac{\gamma + 1}{\gamma - 1} (1 + s(x - ut))^{(\gamma - 1)/2} - \frac{2}{\gamma - 1} \right)$$

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Visualization continued:

$$u = c_0 \left(\frac{\gamma + 1}{\gamma - 1} (1 + s(x - ut))^{(\gamma-1)/2} - \frac{2}{\gamma - 1} \right)$$

Plot $s(x - ut)$ for fixed t , as a function of x :

Let $w = x - ut$

$$x = w + ut = w + u(w)t \equiv x(w, t)$$

$$u(w) = c_0 \left(\frac{\gamma + 1}{\gamma - 1} (1 + s(w))^{(\gamma-1)/2} - \frac{2}{\gamma - 1} \right)$$

Parametric equations:

plot $s(w)$ vs $x(w, t)$ for range of w at each t

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Summary

$$\frac{\partial \rho}{\partial t} + u(\rho) \frac{\partial \rho}{\partial x} = 0$$

Solution: $\rho = \rho_0 + f(x - u(\rho)t) = \rho_0 (1 + s(x - u(\rho)t))$

For linear case: $u(\rho) = c_0$

For non-linear case: $u(\rho) = c_0 \left(\frac{\gamma + 1}{\gamma - 1} (1 + s(x - ut))^{(\gamma-1)/2} - \frac{2}{\gamma - 1} \right)$

Plot $s(x - ut)$ for fixed t , as a function of x :

Let $w = x - ut \Rightarrow x = w + ut = w + u(w)t \equiv x(w, t)$

$$u(w) = c_0 \left(\frac{\gamma + 1}{\gamma - 1} (1 + s(w))^{(\gamma-1)/2} - \frac{2}{\gamma - 1} \right)$$

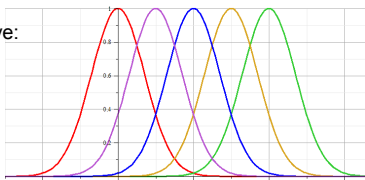
Parametric equations: plot $s(w)$ vs $x(w, t)$ for range of w

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Linear wave:



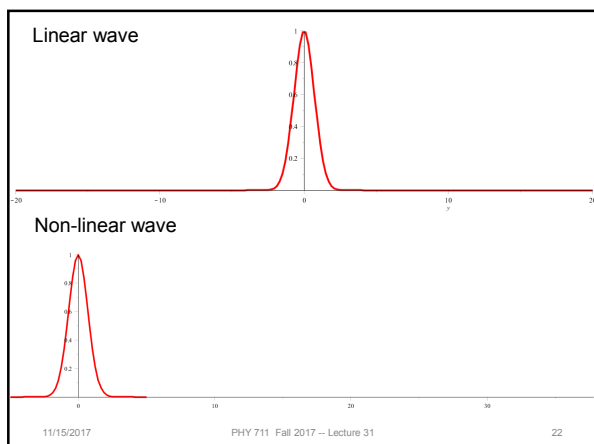
Non-linear wave:

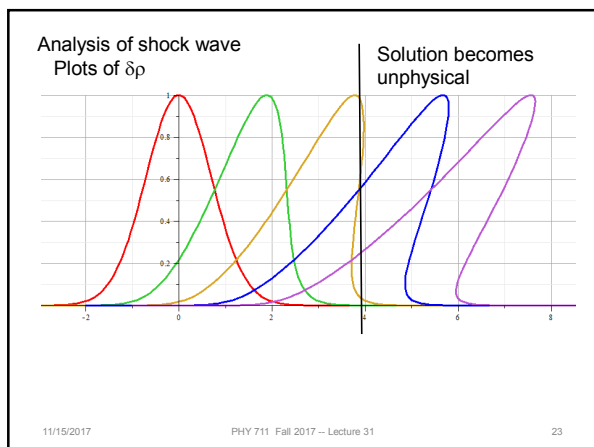


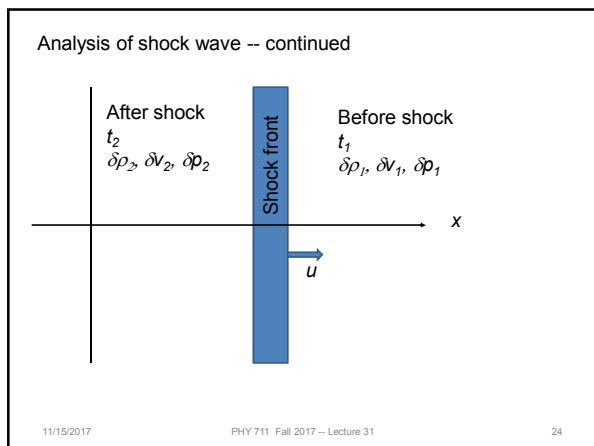
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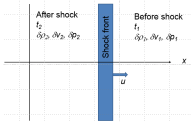






Analysis of shock wave – continued
 While analysis in the shock region is complicated, we can use conservation laws to analyze regions 1 and 2

Assume $\rho(x,t) = \rho(x-ut)$
 $p(x,t) = p(x-ut)$
 $v(x,t) = v(x-ut)$

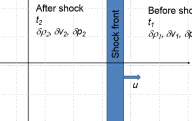


Continuity equation:
 $\frac{\partial \rho}{\partial t} + \frac{\partial(\rho v)}{\partial x} = 0 = \frac{\partial(\rho v - \rho u)}{\partial x} \Rightarrow (v_2 - u)\rho_2 = (v_1 - u)\rho_1$

Conservation of energy and momentum:
 $\Rightarrow p_2 + \rho_2(v_2 - u)^2 = p_1 + \rho_1(v_1 - u)^2$
 $\Rightarrow \epsilon_2 + \frac{1}{2}(v_2 - u)^2 + \frac{p_2}{\rho_2} = \epsilon_1 + \frac{1}{2}(v_1 - u)^2 + \frac{p_1}{\rho_1}$

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Analysis of shock wave – continued
 For adiabatic ideal gas, also considering energy and momentum conservation:



$$\frac{\rho_2}{\rho_1} = \frac{\frac{\gamma+1}{\gamma-1} \frac{p_2}{p_1} + 1}{\frac{\gamma+1}{\gamma-1} + \frac{p_2}{p_1}} \leq \frac{\gamma+1}{\gamma-1}$$

Velocity relationships:
 $\frac{(v_1 - u)^2}{c_1^2} = \frac{1}{2\gamma} \left(\gamma - 1 + (\gamma + 1) \frac{p_2}{p_1} \right)$ $\frac{(v_2 - u)^2}{c_2^2} = \frac{1}{2\gamma} \left(\gamma - 1 + (\gamma + 1) \frac{p_1}{p_2} \right)$
 where $c_1^2 \equiv \frac{\gamma p_1}{\rho_1}$ and $c_2^2 \equiv \frac{\gamma p_2}{\rho_2}$

For a strong shock:
 $\frac{(v_1 - u)^2}{c_1^2} \rightarrow \frac{(\gamma + 1) p_2}{2\gamma p_1}$ $\frac{(v_2 - u)^2}{c_2^2} \rightarrow \frac{(\gamma - 1)}{2\gamma}$

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Analysis of shock wave – continued
 For adiabatic ideal gas, entropy considerations::

Internal energy density: $\epsilon = \frac{p}{(\gamma - 1)\rho} = C_v T$

First law of thermo: $d\epsilon = T ds - p d\left(\frac{1}{\rho}\right)$

$$ds = \frac{1}{T} \left(d\left(\frac{p}{(\gamma - 1)\rho}\right) - p d\left(\frac{1}{\rho}\right) \right) = C_v d \ln\left(\frac{p}{\rho^\gamma}\right)$$

$$s = C_v \ln\left(\frac{p}{\rho^\gamma}\right) + (\text{constant})$$

$$s_2 - s_1 = C_v \ln\left(\frac{p_2}{p_1} \left(\frac{\rho_1}{\rho_2}\right)^\gamma\right) \quad 0 < s_2 - s_1 < C_v \left(\ln\left(\frac{p_2}{p_1}\right) - \gamma \ln\left(\frac{\gamma + 1}{\gamma - 1}\right) \right)$$

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